



Micro Space Propulsion Program

MSP Industry Day John D. Evans September 21, 2007





MSP Goals / Milestones



Program Goal - similar to MEP

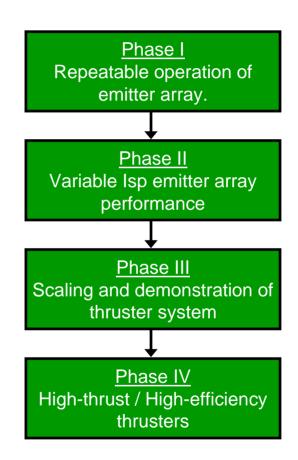
 Develop <u>thruster technology</u> that is <u>scalable</u> to provide <u>on-the-fly control over specific impulse</u>, enabling spacecraft that responsively meet changing national needs.

Then, what is new exactly?

- Change focus to Isp dynamic range.
- Focus on unit cell before scaling

Why a whole new BAA?

- Enable teams to re-scope the problem and form appropriate teams to address our new understanding of technical challenges.
- Bring in more project management expertise.
- Focus earlier on basic emitter performance.
- Refocus effort on Isp dynamic range.





MEP vs. MSP Structure



	Phase I	Phase II	Phase III	Phase IV	
Challenge	Unit Cell		System demonstration		
	Operation	Performance	Scaling		
Duration	TBD*	TBD*	TBD*	TBD*	
Emitters	25+	25+	As needed	As needed	
Power	N/A	N/A	1 W	100 W	
Isp Range	Repeatable** emission >2000 s Repeatable** emission 500 s	Error in BAA	Error in BAA 500s – 10,000 s (on-the-fly adjustable)		
Efficiency***	10%	50%	70%	90%****	

^{*} Phase duration to be determined by offerors, and will be used as a section criterion.

**** Depending on approach, 90% efficiency may not be theoretically possible at 500 s Isp. Offerors should define the efficiency they expect to obtain, and seek to approach theoretical efficiency, to the extent possible.



^{**} For example, >100 on-off cycles, >16 hours of operation, >10 exposures to oxygen containing environment.

^{***} Electrical thrust efficiency, η_t , is defined as: $\eta_t = (FI_sg_o)/2P_e$, where F is the thrust generated by the thruster, Is is the specific impulse, go is the acceleration of gravity, and Pe is the wall-plug power input to the thruster (see for example Sutton, G.P., and Biblarz, O., Rocket Propulsion Elements, John Wiley and Sons, New York, NY, 2001, pg. 665).



MEP vs. MSP Structure



	MEP				
Metric	Phase I	Phase II	Phase III		
Challenge	Individual thruster	Thruster system	Thruster system		
Duration	12 mo.	12 mo.	12 mo.		
Power	1W	100 W	1 kW		
Isp Range	~2500 s, ~7000 s	2000 s - 10,000 s	500 s - 10,000 s		
Efficiency	50%	> 90%	> 90%		

	MSP				
Metric	Phase I	Phase II	Phase III	Phase IV	
Challenge	Unit Cell		Cooling		
	Operation	Performance	Scaling		
Duration	TBD*	TBD*	TBD*	TBD*	
Power	25+ emitters	25+ emitters	1 W	100 W	
Isp Range	Repeatable**	500s - 10,000 s			
	emission < 2000 s				
	Repeatable**	(on-the-fly adjustable)			
	emission 500 s				
Efficiency	10%	50%	70%	90%***	

^{*} Phase duration to be determined by bidders. Used as selection criterion.



^{**}For example , >100 on-off cycles (100x), >16 hrs. operation (100x), >10 oxygen exposure cycles (10x). Well characterized and repeatable performance. Not all emitters need fire in Phase IIa.

^{***90%} emission at 500 s Isp not theoretically possible. Target ~80%.